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An analysis is presented of the tempeature field for a permeable (porous or perforated) wall for a plate, cylinder, and sphere under boundary conditions of the second kind.

Systems consisting of a permeable wall through which an injectant (liquid or gas) is filtered are used extensively in the machine construction and chemical industries as well as in energetics. Despite the urgency of the problem of an analytical analysis of the internal heat and mass transfer in permeable media, this question is studied inadequately in the technical literature [1-6]. The author obtained a solution of this problem in [1-4] for boundary conditions of the first and third kinds. Design dependences for boundary conditions of the second kind are presented below for the following formulation of the problem.

An injectant with the initial temperature T_{ε} and density j_1 of the transverse flux of material is filtered through a wall with porosity Π from the "cold" to the hot surface in the presence of internal energy sources or sinks with the specific power qy(y). The temperatures on the "cold" $(y = y_1)$ and the "hot" surface for $y = y_2$ are T_1 and T_2 , respectively. The thermal flux density at $y = y_2$ is $Q_{2\Gamma}$. It is necessary to find the temperature field in the permeable wall (plane, cylindrical, or spherical) under phase transformation conditions on the "hot" surface of a hood with the influence of all the fundamental process parameters taken into account.

The differential equations for the temperatures of the permeable wall and the injectant, whose derivation and foundation are presented in [1, 6] in the dimensionless variables t = T/T_{∞} and $\bar{y} = y/y_2$, have the form (here and henceforth the prime denotes the derivative with respect to \bar{y})*

$$\bar{y}^{r} t_{r}' + (\Gamma \bar{y}^{r-1} - \xi_{r}) t_{r}' + Q_{r} = 0, \qquad (1)$$

$$\bar{y}^{r} t_{ri}' + (\Gamma \bar{y}^{r-1} - \xi_{ri}) t_{ri}' = 0, r = p. c. s.$$
(2)

under the following boundary conditions: $\bar{y} = -\infty$ (plate); $\bar{y} = 0$ (cylinder and sphere),

$$t_i = t_e, \tag{3}$$

$$\overline{y} = \overline{y}_{1}, \quad t_{r} = t_{r1}, \tag{4}$$

$$\lambda_{ir} t'_i (\bar{y}_i) = q_{ir}, \tag{5}$$

$$\overline{y} = 1, \quad t_{\rm F} = t_{\rm F2}, \tag{6}$$

$$L_{\rm r} + q_{\rm 2r} = \overline{q}_{\rm 2r}.\tag{7}$$

Here

$$q_{\mathbf{i}\mathbf{r}} = \dot{t_{\mathbf{r}}}(\bar{y_{\mathbf{i}}}), \quad q_{\mathbf{2r}} = \dot{t_{\mathbf{r}}}(1),$$

*The subscript Γ indicates what body (one of the three being investigated) is considered, $\Gamma = p$, c, s. If Γ is a factor or exponent, then it will be 0.1 and 2, respectively, for the plate, cylinder, and sphere, where $2\pi\Gamma = 1$ for the plate.

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$$\lambda_{\mathbf{i}\Sigma} = \frac{\lambda_{\mathbf{i}}}{\lambda_{\Sigma}}, \qquad L_{\Gamma} = \frac{rj_{\Gamma}y_{2}^{1-\Gamma}}{\lambda_{\Sigma}T_{\infty}}, \qquad Q_{\Gamma} = \frac{q_{V}(y)y^{\Gamma}}{\lambda_{\Sigma}T_{\infty}y_{2}^{-2}}$$
$$\overline{q}_{2\Gamma} = \frac{q_{2\Gamma}y_{2}}{\lambda_{\Sigma}T_{\infty}}, \qquad \xi_{\Gamma} = \frac{j_{\Gamma}c_{P\mathbf{i}}}{\lambda_{\Sigma}y_{2}^{\Gamma-1}}, \qquad \xi_{\Gamma\mathbf{i}} = \frac{j_{\Gamma}c_{P\mathbf{i}}}{\lambda_{\mathbf{i}}y_{2}^{\Gamma-1}},$$

and $j_{\Gamma} = j_1(\bar{y}_1)y_1^{\Gamma}$, r is the heat of the phase transformations.

Condition (5) characterizes the equality of the conductive thermal fluxes from the injectant and the "cold" wall surface. According to condition (7), the thermal flux $q_{2\Gamma}$ delivered to the permeable wall is expended in the phase transformations and heating of the body skeleton.

Let us first find the solution to the problem (2)-(4). It has the form

$$t_{pi} = (t_{1} - t_{\varepsilon}) \exp \xi_{pi} (\overline{y} - \overline{y}_{i}) + t_{\varepsilon}, \quad -\infty \leqslant \overline{y} \leqslant \overline{y}_{i},$$

$$t_{ci} = (t_{1} - t_{\varepsilon}) \left(\frac{\overline{y}}{\overline{y}_{1}}\right)^{\frac{1}{5}ci} + t_{\varepsilon}, \quad 0 \leqslant \overline{y} \leqslant \overline{y}_{i},$$

$$t_{si} = (t_{1} - t_{\varepsilon}) \exp \xi_{si} \left(\frac{1}{\overline{y}_{1}} - \frac{1}{\overline{y}}\right) + t_{\varepsilon}, \quad 0 \leqslant \overline{y} \leqslant \overline{y}_{i}.$$
(8)

The solution of the differential equation for the permeable body temperature (1) under the boundary conditions (4) and (7) is obtained in two quadratures:

$$t_{\rm r} = t_{\rm ir} + \frac{1}{\xi_{\rm r}} \int_{y_{\rm i}}^{y} Q_{\rm r}(\bar{y}) \, d\bar{y} - \frac{Z_{\rm r}(\bar{y})}{\xi_{\rm r}} + \frac{1}{\xi_{\rm r}} \left[\bar{q}_{2\rm r} - L_{\rm r} + Z_{\rm r}(1)\right] \left[\psi_{\rm r}(\bar{y}) - \psi_{\rm r}(\bar{y}_{\rm i})\right], \quad r = \rm p, \, c, \, s, \tag{9}$$

where

$$Z_{\mathbf{p}}(\bar{y}) = \exp \xi_{\mathbf{p}} \bar{y} \int_{\bar{y}_{1}}^{\bar{y}} Q_{\mathbf{p}}(\bar{y}) \exp \left(-\xi_{\mathbf{p}} \bar{y}\right) d\bar{y},$$

$$Z_{\mathbf{c}}(\bar{y}) = \bar{y} \, {}^{\xi} \mathbf{c} \int_{\bar{y}_{1}}^{\bar{y}} \bar{y}^{-\xi} \mathbf{c} Q_{\mathbf{c}}(\bar{y}) d\bar{y},$$

$$Z_{\mathbf{s}}(\bar{y}) = \exp \left(-\xi_{\mathbf{s}} \, \bar{y}\right) \int_{\bar{y}_{1}}^{\bar{y}} Q_{\mathbf{s}}(\bar{y}) \exp \left(\frac{\xi_{\mathbf{s}}}{\bar{y}}\right) d\bar{y},$$
(10)

$$\psi_{\mathbf{p}}(\overline{y}) = \exp \xi_{\mathbf{p}}(\overline{y} - 1), \ \psi_{\mathbf{c}}(\overline{y}) = \overline{y} \ \xi_{\mathbf{c}}, \ \psi_{\mathbf{s}}(\overline{y}) = \exp \xi_{\mathbf{s}} \left(1 - \frac{1}{\overline{y}}\right),$$

$$Z_{\mathbf{r}}(1) = Z_{\mathbf{r}}(\overline{y})|_{\overline{y}=1}, \ \psi_{\mathbf{r}}(\overline{y}) = \psi_{\mathbf{r}}(\overline{y})|_{\overline{y}=\overline{y}}.$$
(11)

For $Q_{\Gamma} = \gamma_{\Gamma} = \text{const formulas}$ (9) transform into the simpler form

$$t_{\mathbf{r}} = t_{\mathbf{i}\mathbf{r}} + \xi_{\mathbf{r}}^{-1} \gamma_{\mathbf{r}} \left(\overline{y} - \overline{y}_{\mathbf{i}} \right) - \xi_{\mathbf{r}}^{-1} \overline{Z}_{\mathbf{r}} \left(\overline{y} \right) + \xi_{\mathbf{r}}^{-1} \left[\overline{q}_{2} - L_{\mathbf{r}} + \overline{Z}_{\mathbf{r}}(1) \right] \left[\psi_{\mathbf{r}} \left(\overline{y} \right) - \psi_{\mathbf{r}} \left(\overline{y}_{\mathbf{i}} \right) \right].$$
(12)

Here

$$\overline{Z}_{\mathbf{c}} \quad \overline{y} = \frac{\gamma_{\mathbf{c}} \overline{y}^{\,\xi} \mathbf{c}}{1 - \xi_{\mathbf{c}}} \quad \overline{y}^{1 - \xi_{\mathbf{c}}} - \overline{y}^{1 - \xi_{\mathbf{c}}}, \tag{13}$$

$$\overline{Z}_{\mathbf{s}} \quad \overline{y} = \gamma_{\mathbf{s}} \mathbf{H} \quad \overline{y}, \quad \overline{y}_{\mathbf{1}} \exp \left(-\xi_{\mathbf{s}} \overline{y}\right),$$

where

$$\mathsf{M}(\bar{y}, \bar{y}_{i}) = \int_{\bar{y}_{1}}^{\bar{y}} \exp \frac{\xi_{s}}{\bar{y}} d\bar{y}.$$

 $\overline{Z}_{p}(\overline{y}) = \xi_{p}^{-1} \gamma_{p} [\exp \xi_{p}(\overline{y} - \overline{y}_{1}) - 1],$

Differentiating the solutions (8) and (9) in conformity with the boundary condition (5), we find the value of the temperature $t_{1\Gamma}$ on the wall "cold" surface which enters into (9):

TABLE 1. Influence of the Dimensionless Blowing Parameter ξ_{Γ} on the Distribution of the Temperature t_{Γ} over the Porous Wall Thickness y and Values of $q_{1\Gamma}$ for $q_{2\Gamma}$ = 0.999

			Ţ.										
Г	÷,	<i>q</i> 10	0,80	0,82	0.84	0.86	0,88	0,90	0.92	0,94	0.96	0,98	1,00
			t _r ·10 ^a										
	2	1.49	847	876	905	933	960	986	1010	1030	1060	1080	1100
	3	1.30	533	559	585	£09	634	657	680	703	725	746	766
0	5	1.00	300	320	340	360	380	400	420	440	460	480	500
	7	0.784	212	228	244	260	277	249	311	329	347	366	386
	10	0,568	157	168	180	192	205	218	232	247	263	281	300
	2	1.80	820	855	888	920	951	980	1010	1030	1060	1080	1100
	3	1,54	510	541	570	598	626	652	677	701	724	746	766
1	5	1,15	284	306	329	351	374	395	417	438	459	480	500
	7	0,887	200	218	236	253	272	290	308	327	346	366	386
	10	0,615	149	162	174	188	201	215	230	246	263	281	300
	2	2,15	789	831	870	i S07	941	973	1000	1030	1060	1080	1100
2	3	1,81	485	521	555	587	617	646	673	699	723	746	766
	5	1,20	266	292	317	342	367	391	414	436	458	479	500
	7	0,964	188	207	227	247	266	286	306	326	346	366	386
	10	0,655	142	155	169	183	197	212	228	245	262	280	300

TABLE 2. Dependence of t_{Γ} on \bar{y} for Different Values of the Parameter L_{Γ}

	L _r	9 ₁₁	<i>q</i> _{2Г}	<u> </u>										
Г				0,80	0,82	0,84	0,86	0,88	0.90	0,92	0,94	0,96	0,98	1,00
				t _r .10*										
0 1 2	0,011 0,051 0,081 0,011 0,051 0,081 0,011 0,051 0,081	1,13 1,11 1,10 1,32 1,30 1,28 1,52 1,50 1,48	0,989 0,949 0,919 0,989 0,949 0,919 0,989 0,949 0,949 0,919	383 379 375 364 360 357 343 340 337	406 401 397 390 386 382 373 369 366	428 423 419 416 411 407 402 398 394	450 445 440 441 435 431 430 425 421	472 466 461 465 459 455 458 452 447	494 487 482 489 482 477 484 477 472	515 508 502 512 505 499 509 509 502 496	536 528 522 534 527 521 533 525 519	557 548 542 556 548 541 555 547 540	577 568 561 577 568 561 577 568 561	597 587 580 597 587 580 597 587 587 580

$$t_{ir} = t_e + p_c \psi_r(\overline{y}_i), \quad r = p, c, s.$$
(14)

For $Q_{\Gamma} = \gamma_{\Gamma} = \text{const}$ we obtain from (14)

$$t_{ir} = t_e + \overline{\rho_r} \psi_r (\overline{y_i}).$$

Here

$$p_{\rm r} = \xi_{\rm r}^{-1} \left[\overline{q}_{2\rm r} - L_{\rm r} + \gamma_{\rm r} \varkappa_{\rm r} \left(\overline{y}_{\rm i} \right) \right], \tag{15}$$

where

$$\varkappa_{p}(y_{1}) = \xi_{p}^{-1} [\exp \xi_{p}(1 - y_{1}) - 1],$$

$$\varkappa_{c}(\bar{y}_{1}) = \frac{1 - \bar{y}_{1}^{\xi_{c}}}{1 - \xi_{c}},$$

$$\varkappa_{s}(\bar{y}_{1}) = \mathcal{H}(1, \bar{y}_{1}) \exp(-\xi_{s}).$$
(16)

The absolute values of the thermal fluxes $q_{1\Gamma}$ and $q_{2\Gamma}$ in the boundary conditions (5) and (7) are found by differentiating the solution (9) with (14) taken into account:

$$q_{ir} = \bar{y}_{1}^{-r} [\bar{q}_{2r} - L_{r} + Z_{r}(1)] \psi_{r}(\bar{y}_{i}), \qquad (17)$$

$$q_{2r} = \overline{q}_{2r} - L_r, \tag{18}$$

where (18) agrees with (7).

For $Q_{\Gamma} = \gamma_{\Gamma} = \text{const}$ we obtain from (17)

 $q_{\mathbf{ir}} = \overline{y}_{\mathbf{l}}^{-r} \left[\overline{q}_{\mathbf{2r}} - L_{\mathbf{r}} + \overline{Z}_{\mathbf{r}} \left(\mathbf{l} \right) \right] \psi_{\mathbf{r}} \left(\overline{y}_{\mathbf{l}} \right).$

The results of computing the dimensionless temperature of the permeable wall t_{Γ} and the thermal fluxes are presented in Tables 1 and 2, as well as in Figs. 1-3. The data presented



Fig. 1. Distribution of the dimensionless temperature t over the wall thickness $l = 1 - \bar{y}_1$, divided into n parts, for $\bar{y} = 0$ and $\bar{y} = 1$, respectively, n = 0 and n = 10 [1) $\bar{y}_1 = 0.2$; 2) 0.3; 3) 0.4; 4) 0.5; 5) 0.6; 6) 0.7; 7) 0.8; 8) 0.9]. Here and in Figs. 2 and 3: a for a plate; b) for a cylinder; c) for a sphere.



Fig. 2. Distribution of the temperature t over the wall thickness y as a function of the dimensionless thermal flux $\bar{q}_{2\Gamma}$ to the "hot" surface: 1) $\bar{q}_{2\Gamma} = 5.0$; 2) 2.0; 3) 0.7; 4) 0.4; 5) 0.1; 6) 0.

have been obtained for $\xi_{\Gamma} = 4$, $\gamma_{\Gamma} = 5$, $L_{\Gamma} = 0.001$, $q_{2\Gamma} = 1$. One of these parameters is assumed variable in order to analyze its influence on the process under consideration.

The data obtained confirm the fact that the values of the temperature t_{Γ} and their gradients over the wall thickness diminish as the dimensionless blowing velocity ξ_{Γ} increases. The $q_{1\Gamma}$ diminish correspondingly; hence the values of $q_{2\Gamma}$ remain constant since the computation was performed for $\overline{q}_{2\Gamma} = 1$. As the heat expenditure in the phase transformations, characterized by the parameter L_{Γ} (Table 2), increases, the values of $t_{1\Gamma}$, $q_{1\Gamma}$, and $q_{2\Gamma}$ diminish, which is explained by condition (7). As the wall thickness $l = y_2 - y_1$ increases in proportion to the number n (n = 0 and n = 10, respectively, for y = y_1 and y = y_2), the values of the dimensionless temperatures t_{Γ} and the differences $\Delta t = t_2 - t_1$ grow (Fig. 1). As should be expected, the values of t_{Γ} also rise with the growth of the flux $\overline{q}_{2\Gamma}$ delivered to the wall hot surface (Fig. 2), as well as the energy source intensity γ_{Γ} (Fig. 3).

The greatest values of the temperature t_{Γ} and the differences $\Delta t_{\Gamma} = t_{2\Gamma} - t_{1\Gamma}$ are observed for a sphere and the least for a plate in all the cases considered, which is explained by the following relationship governing the change in the weight discharge of the injectant over the wall thickness of the porous body:



Fig. 3. Dependence of t on the coordinate y and internal energy intensity γ_{Γ} : 1) $\gamma_{\Gamma} = 10$; 2) 7; 3) 3; 4) 0; 5) -3.

TABLE 3. Analysis of Parameters ξ_{Γ} , $\overline{q}_{2\Gamma}$ for λ_{Σ} = 0.418 W/m·deg

Gas being filtered	/ j. 10 ³ . kg/m ² · sec	^c ⊳i, kJ/kg•deg		ē _r	4 2F	
Na tura l	5,83	1,25	1070	1,75	4,6	
	1,80	1,25	900	0,53	1,1	
Coke	3,62	3,56	1000	3,07	8,5	
Liquefied	3,33	4,18	1140	4,0	9,7	

$$\dot{j}_{i}(\bar{y}) = \dot{j}_{i}(\bar{y}_{i}) \left(\frac{y_{i}}{y}\right)^{r},$$

where $j_i(\bar{y}_1)$ is the value of j_i at the "cold" surface of the permeable wall.

The dependences (9) and (12) have been obtained under the assumption of constant thermophysical properties. As the analysis performed showed, the error in the analysis because of such an assumption is around 10%, which corresponds to experimental accuracy [6].

Values of the dimensionless parameters ξ_{Γ} and $\bar{q}_{2\Gamma}$ computed at $T_{\epsilon} = 293^{\circ}K$, $y_2 = 0.1$ are presented in Table 3 according to test data obtained for a porous ceramic radiator [7].

The values of L_{Γ} used in the computations correspond to the actual densities of a transverse flux of liquid injectant j_i which evaporates on the body surface. Thus, in the case of water evaporation $j_i = 0.24 \cdot 10^{-4}$ and $j_i = 1.76 \cdot 10^{-4} \text{ kg/m}^2 \cdot \text{sec}$, we have $L_{\Gamma} = 0.011$ and $L_{\Gamma} = 0.081$, respectively. Corresponding to the value $q_V = 61.25 \text{ kW/m}^3$ is $\gamma_{\Gamma} = 5$. Therefore, the considered values of the dimensionless parameters correspond to the actual range of their variation under practical conditions. In the general case, the solutions obtained are valid for values of L_{Γ} , Q_{Γ} , and ξ_{Γ} which vary between 0 and ∞ [1].

If $\bar{q}_{2\Gamma} = (1 - t_2)\alpha y_2/\lambda_{\Sigma}$, where α is the heat transfer coefficient, then the solutions under the boundary conditions of the second and third kinds agree [1]. In contrast to the boundary conditions of the first kind, the heat balance equation for $\bar{y} = 1$ hence permits finding t_2 and taking account of the influence of energy sources or sinks on this surface.

NOTATION

T, permeable wall temperature; Π , its porosity; y, a coordinate normal to the surface of the permeable body; λ , heat conduction coefficient; $\lambda_{\Sigma} = (1 - \Pi)^{\lambda} T + \Pi \lambda_{L}$. Subscripts: T, porous body skeleton; i, injectant; Σ , total (effective) value; 1, "cold" wall surface; 2, "hot" wall surface, ε , values as $y \to -\infty$, ∞ as $y \to +\infty$; p, plate; c, cylinder; and s, sphere.

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OPTIMIZATION OF CONVECTIVE CIRCULAR FINS

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The volume of a circular fin whose thickness is inversely proportional to the square of the radius is optimized.

The books [1, 2] provide an idea of the present state of the theory and practical application of finned heating surfaces. These books also examine the question of optimizing the volume of the fins. The object of optimization is to select a fin with minimum volume for transferring a specified amount of heat under known thermophysical conditions. Whereas for straight fins the problem of optimization is solved by several variants of the cross section of the fin, for circular fins only the results of [3] for fins of constant thickness are given.

We note that for hyperbolic profiles examined in [1, Tables 1-5], the problem of optimization is solved very simply in the case of the thickness of the fin being inversely proportional to the square of the radius. If we use the notation of [1], this dependence has the form

$$\delta/\delta_1 = R^{-2},\tag{1}$$

$$R = r/r_{i}.$$
 (2)

For convenience, we denote the height of the fin

$$h = r_2 - r_1, \tag{3}$$

and the parameter of the fin N is expressed in the form

$$N^2 = 2\alpha h^2 / \lambda \delta_1. \tag{4}$$

To make the circular rib more comparable with a straight rib, we refer the thermal flux and the volume of the circular rib to a unit length of the base

$$Q_1 = Q_0/2\pi r_1 = \alpha \vartheta_1 \eta h(R_2 + 1), \tag{5}$$

$$V_{1} = \frac{1}{2\pi r_{1}} \int_{r_{1}}^{r_{2}} 2\pi r \delta dr = h \delta_{1} \ln R_{2} / (R_{2} - 1).$$
(6)

Determining the value of δ_1/h^2 from (4), and h from (5), we can express the product in (6) as

$$h\delta_{\mathbf{i}} = (\delta_{\mathbf{i}}/h^2) (h^3), \tag{7}$$

and formula (6) is transformed to the form

$$V_{i} = \left(\frac{Q_{i}}{\alpha \vartheta_{i}}\right)^{3} \frac{2\alpha}{\lambda N^{2}} \frac{\ln R_{2}}{\eta^{3}(R_{2}+1)^{3}(R_{2}-1)}.$$
(8)

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